

Solar Probe Plus

A NASA Mission to Touch the Sun

Integrated Science Investigation of the Sun Energetic Particles

Preliminary Design Review

05 – 06 NOV 2013

EPI-Lo Calibration

Don Mitchell

EPI-Lo Scientist (JHU/APL)



This document contains technical data that may be controlled by the International Traffic in Arms Regulations (22 CFR 120-130) and may not be provided, disclosed or transferred in any manner to any person, whether in the U.S. or abroad, who is not 1) a citizen of the United States, 2) a lawful permanent resident of the United States, or 3) a protected individual as defined by 8 USC 1324b(a)(3), without the written permission of the United States Department of State.



Outline



- Overview
- Species and Energy Ranges
- Facilities
- Calibration Plan
- Test Flow
- In-Flight Calibration



EPI-Lo Calibration



- EPI-Lo measurements are intended to generate the information needed to derive differential intensities ($j[\text{cm}^2 \cdot \text{sr} \cdot \text{s} \cdot \text{keV}]^{-1}$)
- The goal of EPI-Lo characterization and calibration efforts is to develop the quantitative procedures for converting the count rates ($R [\text{counts s}^{-1}]$) reported by EPI-Lo into estimates of j for the various defined ranges of energies, particle species, and arrival angles
- “Calibration” for a particle instrument like EPI-Lo means determining the following:
 - Transfer function from counts into flux (physical units)
 - Characteristic of “Rate-in” versus “Rate-out”
 - Response to visible and ultraviolet light
 - Response to high energetic particle backgrounds



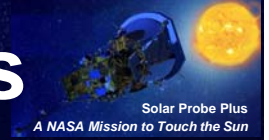
Calibration Types



- Foreground
 - Ions and electrons in the energy range of interest to the instrument
- Background
 - Electrons
 - Characterize the rates from penetrating radiation
 - Characterize response to Solar Wind and/or photoelectron impact
- Light
 - Characterize rejection of UV background, primarily H-alpha
 - Characterize rejection of sunlight and glint



Foreground Calibration Requirements



- Foreground Electrons
 - 40 keV to 1000 keV (Needed for understanding backgrounds)
 - 1 keV to 30 keV Electron Gun at APL
 - 30 keV to 100 keV Radioactive sources at APL
 - 125 keV to 1.6 MeV Accelerator at GSFC
- Foreground Ions (H, He3, He4, O, Fe)
 - 40 keV to 15000 keV (Level 4 Requirements)
 - (Goal: protons to 20 MeV)
 - 3 keV to 170 keV Accelerator at APL
 - 30 keV to 5 MeV Degraded alpha sources
 - 125 keV to 1.6 MeV Accelerator at GSFC
 - 1 MeV to 20 MeV Accelerator at LBL



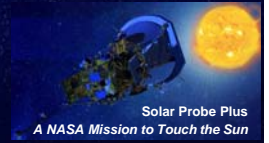
Test as You Fly



- The instruments will be tested in flight-like environments
- Since the instrument will need to operate in a high background environment, we will characterize response to high energy penetrating radiation, UV light, and low energy plasma (all potential sources of background counts for EPI-Lo)



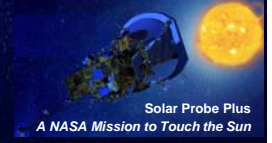
Background Calibration Requirement



- Background Electrons
 - 2 eV to 10 MeV (from the expected environment)
 - 1 eV to 100 eV Hot filament at APL
 - 1 keV to 50 keV Electron Gun at APL
 - 125 keV to 1.6 MeV Accelerator at GSFC
 - 1 MeV to 10 MeV Accelerator at Idaho
- Background Ions
 - 3 keV to 50 MeV (from the expected environment)
 - 3 keV to 170 keV Accelerator at APL
 - 30 keV to 5 MeV Degraded alpha sources
 - 125 keV to 1.6 MeV Accelerator at GSFC
 - 1 MeV to ~100 MeV Accelerator at LBL
- Photons
 - UV and visible lamps at APL



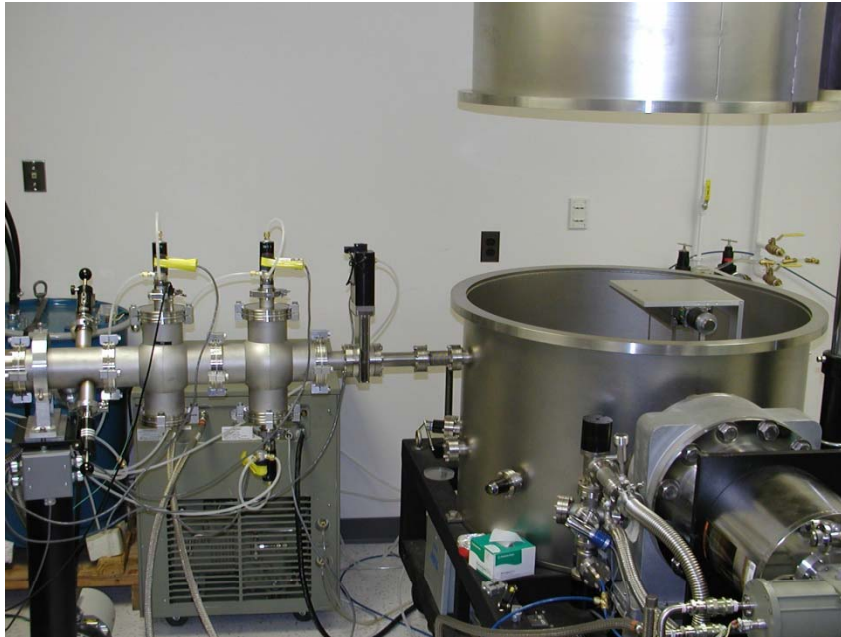
The JHU/APL Calibration Facility



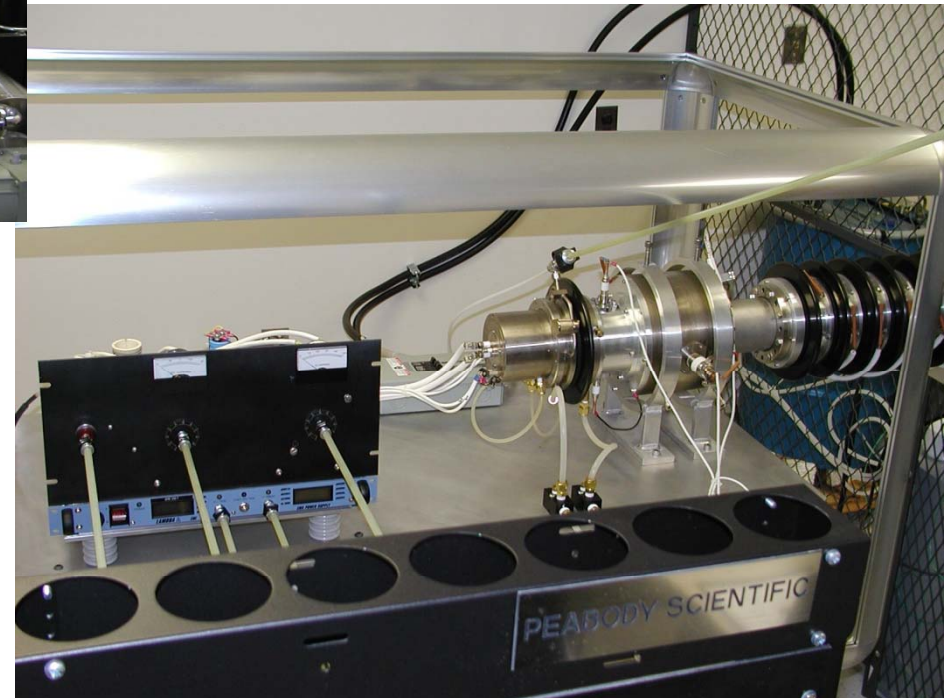
- The APL particle accelerator is a versatile system capable of producing a broad range of ion species at energies from 20 to 170 keV
- The system includes a electron-impact ionization source, extraction gap, Einzel Lens and Wien filter mounted in the insulated terminal structure along with all associated power supplies
- The system will produce beams of H, He, O, and noble gas ions with intensities over the range of 100's to 1,000,000 particles/cm²/sec at the target position (mm² - cm²)
- We also have a variety of radioactive sources as stimulus



The JHU/APL Calibration Facility



- All ions from a gas source
- Energy continuously tunable: 3 to 170 keV
- Wien filter
- Beam intensity between 10 and 10¹⁰ ions/cm²/s
- Purposed built articulation stage





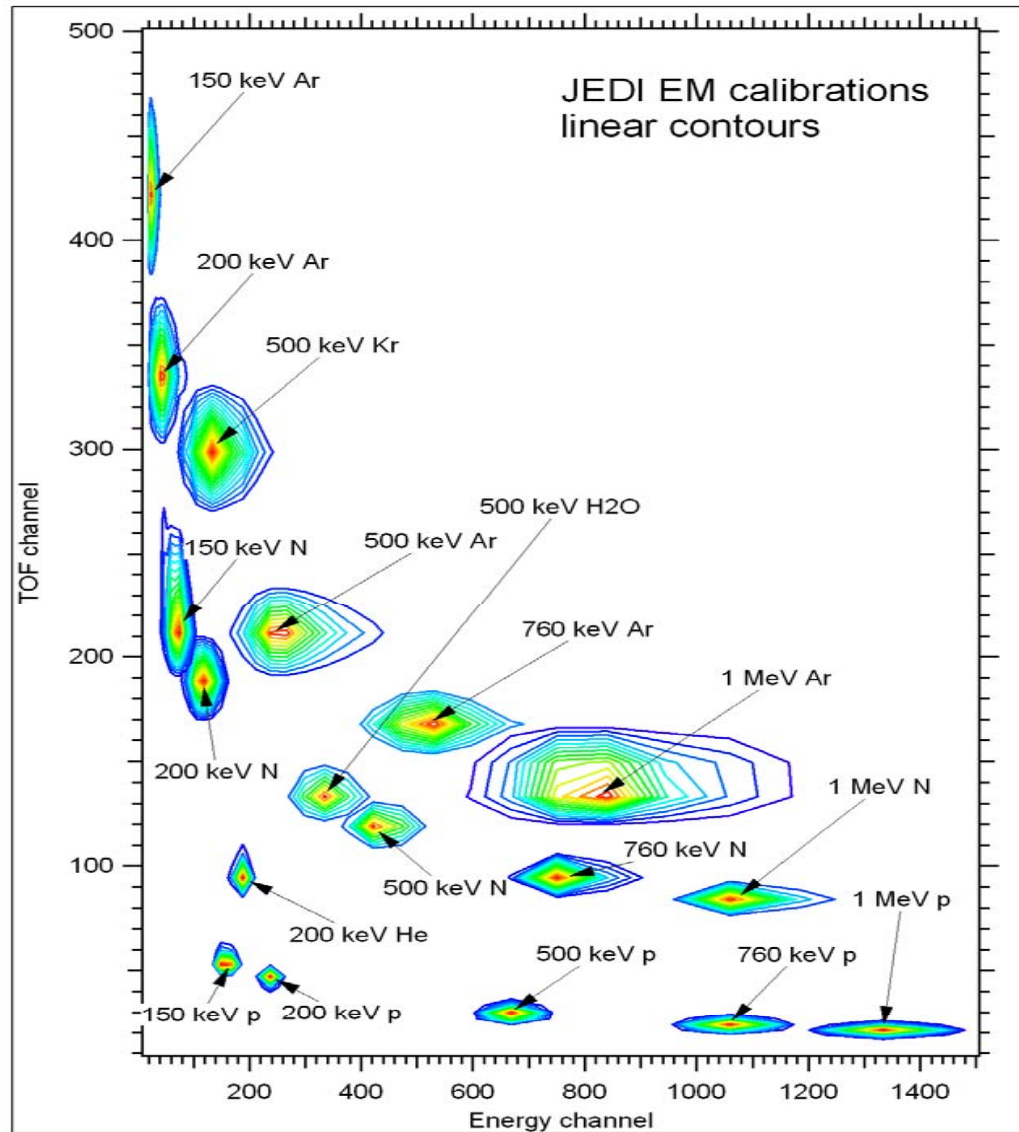
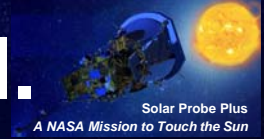
Beam Tests at GSFC



- It is planned to have four calibration sessions at the accelerator at the Goddard Space Flight Center. Each session starts with a one-day setup, check, and pump:
- Session 1: Exploratory run to characterize EPI-Lo
- Session 2: Use H beam to scan both angles to complete characterization of the transfer function
- Session 3: Characterize sensor response with e-beam from ~ 100 keV to 1 MeV
- Session 4: Use heavy ions (He, O, and Ar) to characterize the instrument response



Representative Results from JEDI Cal.





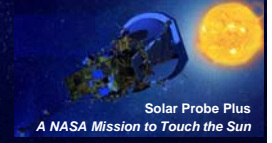
EPI-Lo Test Summary



- Prototype “EM” Testing
 - Validate instrument design and performance
 - Energy response
 - Instrument efficiency
 - Instrument geometry factor
 - Establish testing procedures
- Flight Model (FM)
 - Verify instrument design and performance
 - Energy response
 - Instrument efficiency
 - Instrument geometry factor



Calibration Approach (Flight Units)



- FM Unit
 - All instrument integration activities will be performed in a Class 5 clean room environment
 - Test in bell jar to characterize geometry, energy response, and sensitivity
 - Calibrate using particle sources and in Beam Facilities at APL to characterize energy response, sensitivity, dynamic range
 - Compare with EM results to cover gaps in energy coverage
- Test Philosophy:
 - FM will be extensively calibrated, and performance compared with more extended energy range EM calibrations (LBL, GSFC)
 - Pre- and post-environmental qualification spot calibration
 - In-flight cross calibration between EPI-Lo and EPI-Hi



EPI-Lo Calibration Schedule



- Final calibration for FM slated for three weeks
- Major calibration efforts will be performed at APL facility
 - Substantial understanding of the instrument performance will be made with the EPI-Lo EM characteristics



Tests at Instrument Level



| | Element | Property | Requirement* | Expected Performance | Calibration Accuracy |
|--------------|-------------------------------------|--|--|--|--------------------------|
| System Level | Calibration mode without Collimator | Energy-ToF plane characteristic | Verify simulation to 20% | Verify simulation to 5% | 5% |
| | | Input/output rate at system level | Known to 10% | Known to 2% | 1% |
| | | Background rejection | > 90% | > 95% | 2% |
| | | Mass resolution | Discriminate between e^- , H^+ , $3He^+$, $4He^+$, C , O^+ , | < 0.5 AMU (H^+) < 1 AMU (CNO) < 2 AMU (Fe^+) | 0.5 AMU |
| | | Absolute efficiency | Known to 50% for e^- , H^+ , He^+ , CNO+ | 10% | 10% |
| | | | | | |
| | Calibration mode with collimator | Scattering of ions | < 10% | < 5% | 2% |
| | | Scattering of electrons | < 10% | < 5% | 2% |
| | | Properties at octant boundaries | Known to 30% | Known to 5% | 5% |
| | | Efficiency as a function of entrance | Known to 20% | Known to 5% | 5% |
| | | Angular resolution | 30° | 25° | 3° |
| | | Geometric factor | > 0.05 cm ² -sr | 0.061 cm ² -sr | 0.01 cm ² -sr |
| | | Full calibration: verify previous measurements | | | |
| | Flight mode | Input/output rate at system level | Known to 30% | Known to 10% | 10% |
| | | Verify all modes | | | |
| | | Verify all timing windows | | | |
| | | Throughput of event classification | | | |
| | | Efficiency of counters | | | |
| | | Energy-ToF plane characteristics | | | |
| | | Threshold settings | | | |
| | | Temperature dependent | | | |

*Science requires relative/absolute accuracy: 20%/50%. Ground calibration 20% precision, reduced to 10% in flight.



In-Flight Calibration



- On-orbit and cruise calibration achieves relative calibration to 10% precision
 - Uniformity confirmed by evolution of pitch angle distribution from onset to shock passage.
 - Such calibrations cover the entire energy and FOV coverage with linear instrument response (targeted rates, no pulse pileup)
- Built-in features to determine on-orbit instrument ion performance
 - Measure pulse-height spectrum of secondary electrons from incident protons as function of time-of-flight
 - Unit has built-in stimulus to inject known pulse through the front-end electronics



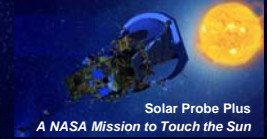
Summary



- Calibration plan satisfies all Level IV requirements
- Calibration facilities have been identified that meet EPI-Lo needs
- APL operates and maintains the key EPI-Lo calibration facility which allows maximum flexibility



Accuracy & Precision



| Parameter | Required | Goal (Capability) | Comment/Heritage |
|--|---|--|--|
| Electron Energies | 50 - 500keV | 25 - 1000 keV | Electron capability from JEDI, RBSPICE |
| Ion Energies | 50 keV/nucleon – 15000 keV Total E | 40 keV/nucleon – 20000 keV Total E | Capability partially based on RBSPICE capabilities. Top energy ~250keV/nuc for Fe |
| Energy Resolution | 45% for required energy range | 40% for required energy range | Telemetry limited |
| Time sampling | 5 sec | 1 sec | Telemetry and/or statistics limited |
| Angle resolution | $<30^\circ \times <30^\circ$ | Ions, $\sim 15^\circ \times 12^\circ$ to $<30^\circ \times <30^\circ$ e-, 45° | Varies with elevation |
| Pitch Angle (PA) Coverage | $0^\circ - 90^\circ$ or $90^\circ - 180^\circ$, some samples in both hemispheres | $0^\circ - 90^\circ$ or $90^\circ - 180^\circ$, some samples in both hemispheres | |
| Time for Full PA | 1 – 5 sec | 1 – 5 sec | Telemetry limited |
| Ion Composition | H, He3, He4, C, O, Ne, Mg, Si, Fe | H, He3, He4, C, O, Ne, Mg, Si, Fe | He3/He4 ~50 to 1000 keV/nuc |
| Electron Sensitivity: I=Intensity ($1/\text{cm}^2\cdot\text{s}\cdot\text{sr}$) | $j = 1\text{E}1\text{-}1\text{E}6/\text{cm}^2\cdot\text{s}\cdot\text{sr}$ | Sensor-G: $0.144 (\text{cm}^2\cdot\text{s}\cdot\text{sr})$ Pixel-G: $\sim 0.02 (\text{cm}^2\cdot\text{s}\cdot\text{sr})$ Up to $6\text{E}6$ 1/s counting | j =Intensity ($1/\text{cm}^2\cdot\text{s}\cdot\text{sr}$) G =Geom. Factor ($\text{cm}^2\cdot\text{s}\cdot\text{sr}$) 8 pixels/sensor |
| Ion Sensitivity | $j = 1\text{E}1\text{-}1\text{E}6/\text{cm}^2\cdot\text{s}\cdot\text{sr}$ | Sensor-G: $0.16 (\text{cm}^2\cdot\text{s}\cdot\text{sr})$ Pixel-G: $\sim 0.002 (\text{cm}^2\cdot\text{s}\cdot\text{sr})$ Up to $3.5\text{E}6$ 1/s counting (TOFxE) | 80 pixels/sensor |